

# Validation of thermo-mechanical deformation of an in-orbit antenna

*Third International Workshop on the Validation of Computational Mechanics  
Models*

Alexander Ihle & Olaf Reichmann

Munich | June 12<sup>th</sup> 2014

## Table of Contents

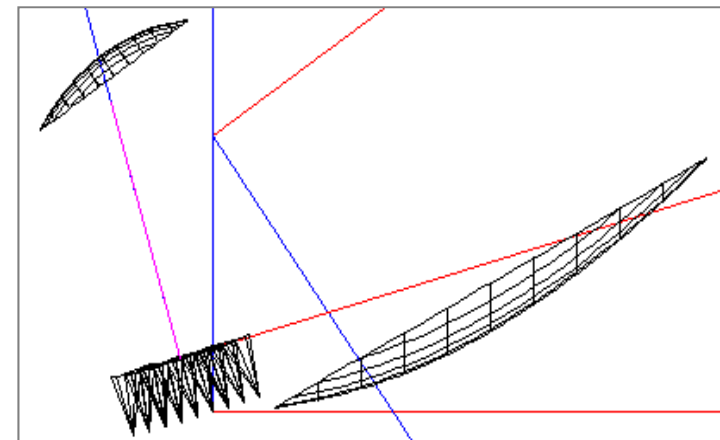
- Part 1: Technical Concept**
- Part 2: ESPI**
- Part 3: STANT Thermo-Mechanical Deformation Test and Modeling**
- Part 4: Conclusion and Outlook**

# Part 1: Technical Concept



## Highly Stable Antenna Technologies (STANT)

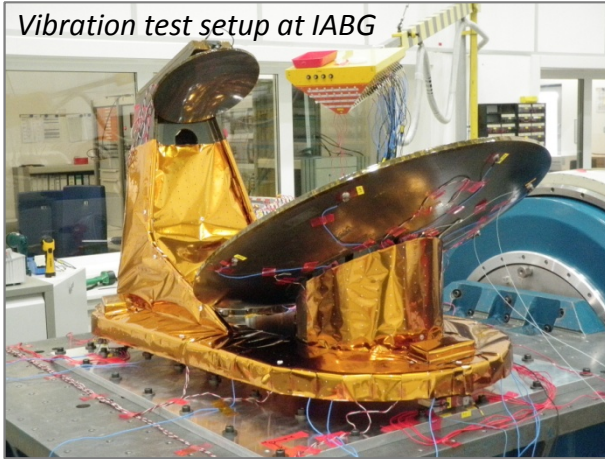
- ESA ARTES 5.2 activity
- Ka-band & Q/V-band
- 1 to 19 RF beams, European coverage
- Compact
- High thermo-elastic stability
- High stiffness to mass ratio
- Low heat transfer between S/C and antenna
- Neither reflector coating nor sunshields
- For CFRP & Alu-satellite top decks
- Mass below 10kg
- 1 eigenfrequency > 100Hz
- In-orbit deformation < 20 $\mu$ m RMS.



*Gregorian dual reflector configuration*

### STANT Testing

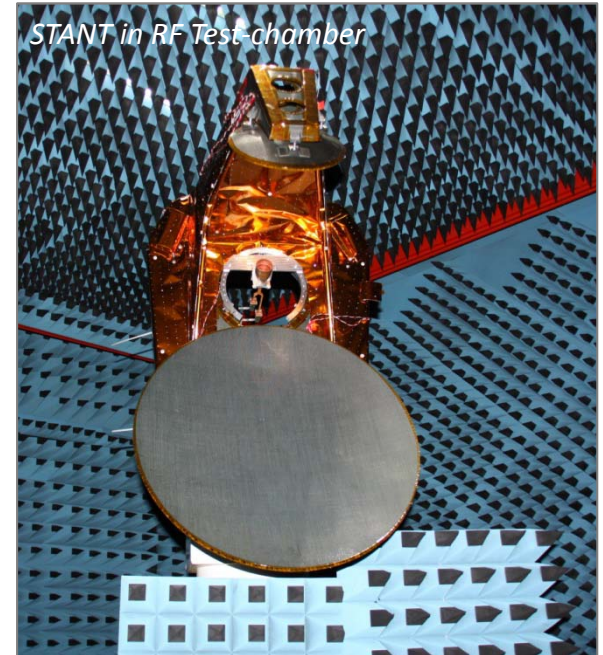
*Vibration test setup at IABG*



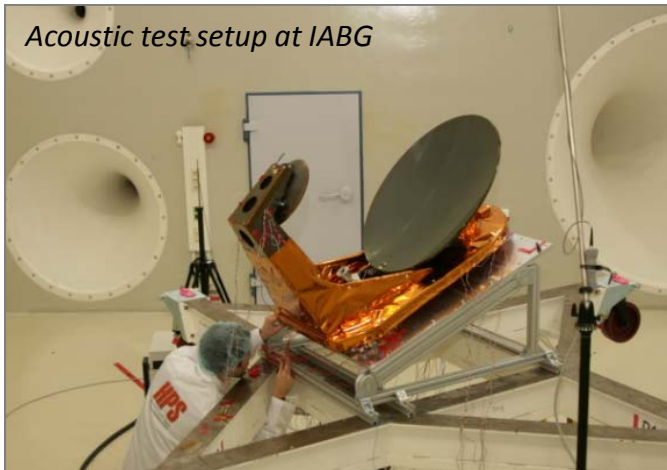
*STANT in TV-chamber*



*STANT in RF Test-chamber*



*Acoustic test setup at IABG*





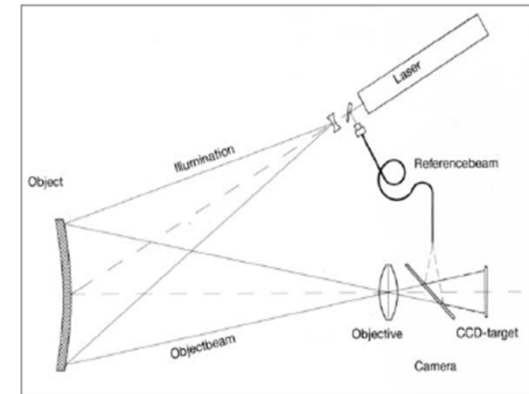
Stable Antenna Technology  
(STAT) Engineering

## Part 2: ESPI

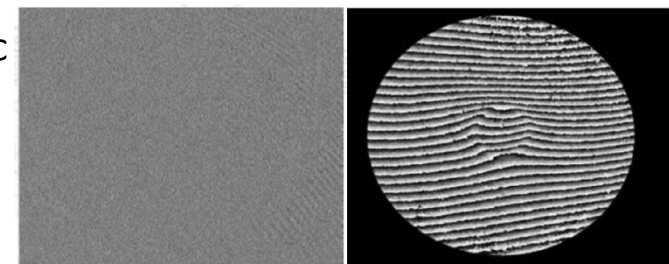


## Electro Speckle Pattern Interferometry

- Optical measurement:
  - Illuminate target with pulsed laser
  - Record back scattered light “Speckle Pattern”
  - Compare with reference beam “PhaseImage”
  - Phase image remains constant as long as geometry between laser, camera and object does not change
- ESPI data analysis:
  - Motion of object causes change of phase image
  - Difference of two phase images includes rigid-body and elastic displacement information of object in terms of interferometric fringes “Interferogram”
  - Unwrapping and post processing of the interferograms separates rigid-body motion from elastic deformation part by best fit calculation for rigid-body part
- Accumulation of many successive interferograms results in total deformation map



ESPI optical path



Speckle pattern

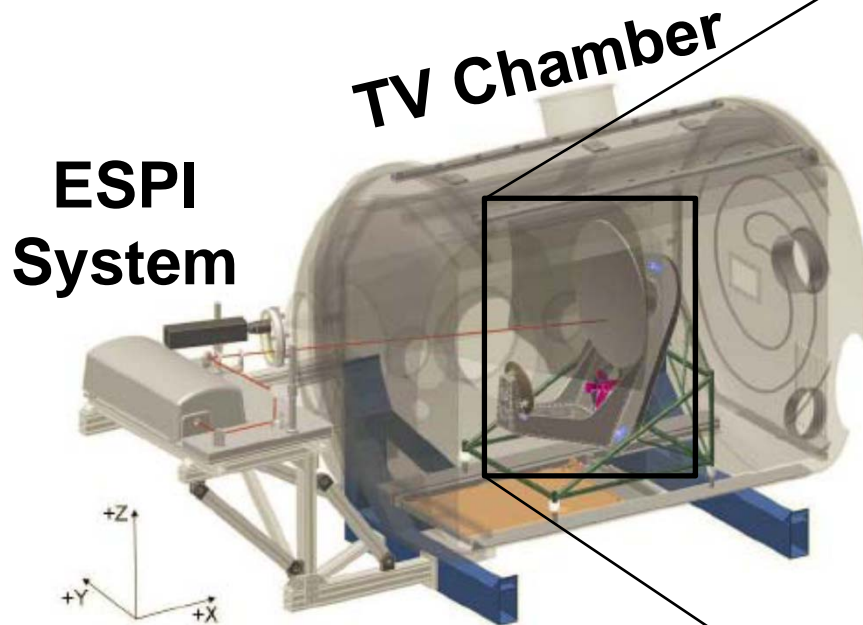
Interferogram

# Part 3: STANT Thermo-Mechanical Deformation Test and Modeling





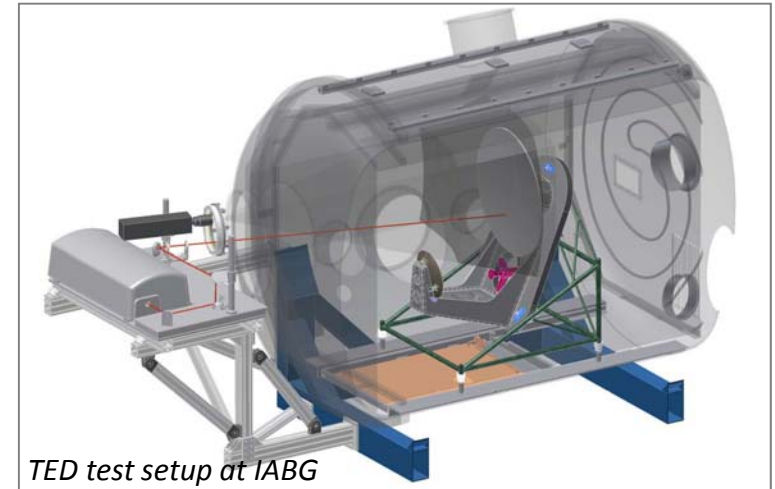
# STANT TED Test Campaign



**STANT Antenna Reflector**

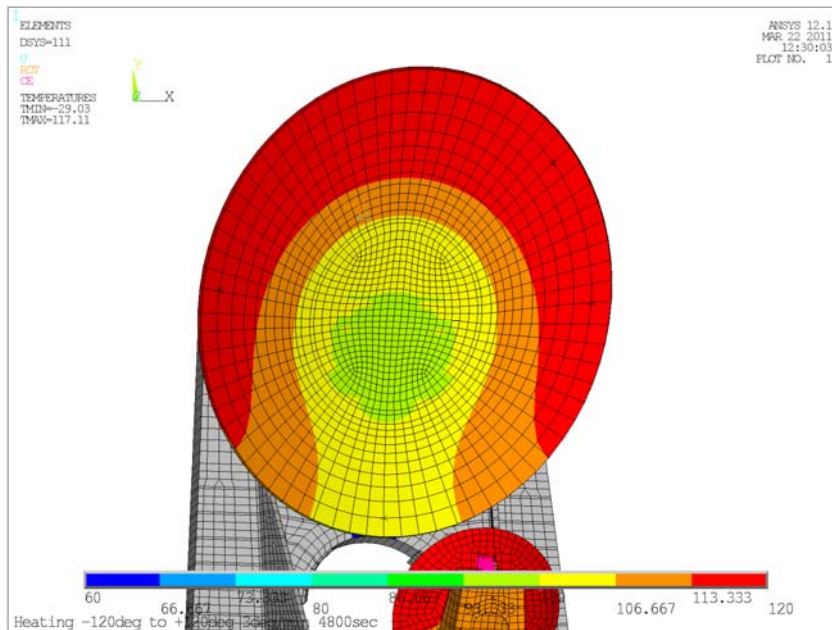
## Thermal Cycling and Thermo-Elastic Test

- 3 in vacuo cycles from -120 C to +120 C
- TED-measurement of main reflector
- Challenges
  - Any rigid-body motion must be limited in magnitude
  - Vibration isolation of the chamber
  - Thermal isolation of fixture from chamber
  - Window (spectral transmissibility for external sensor)
  - Reflector surface (optical contrast at reflector level)
  - Data processing
- Measurement via ESPI
- Results
  - No failure and low TED

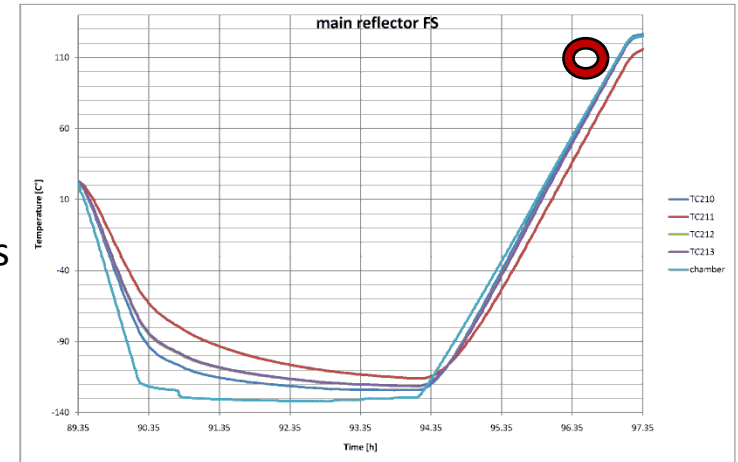


## Thermal Analysis

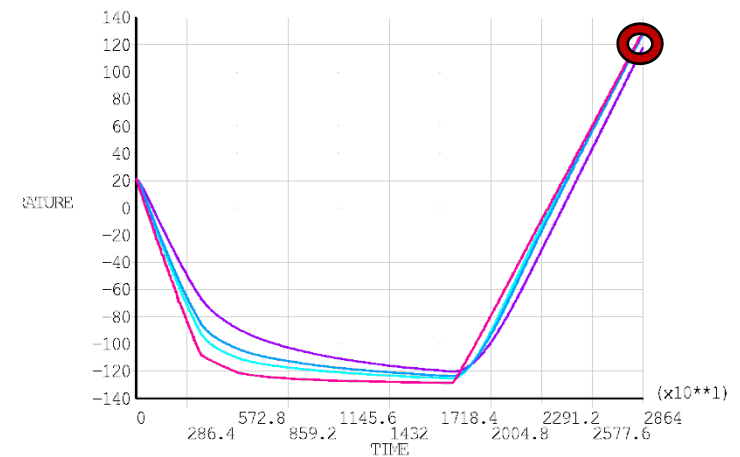
- Correlation of thermal analysis and measured temperatures of TV cycling test
- Mapping of temperature results to structural model and analysis of corresponding deformations



Simulated temperatures on MR at end of cycle 1



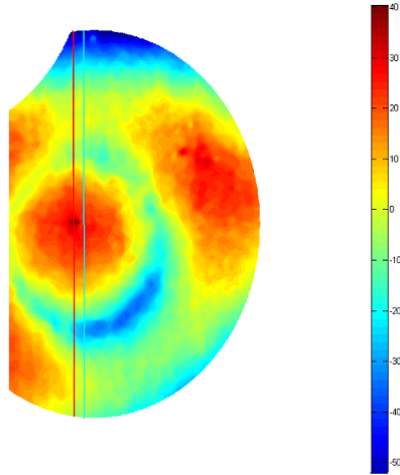
Measured temperatures on MR at cycle 1



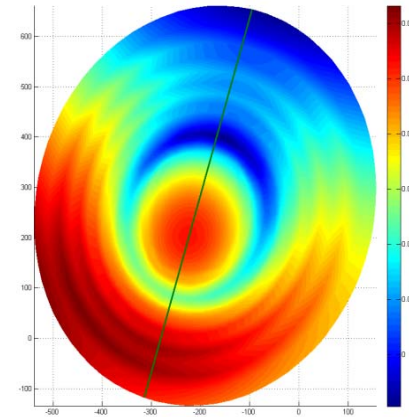
Simulated temperatures on MR at cycle 1

# Initial Correlation

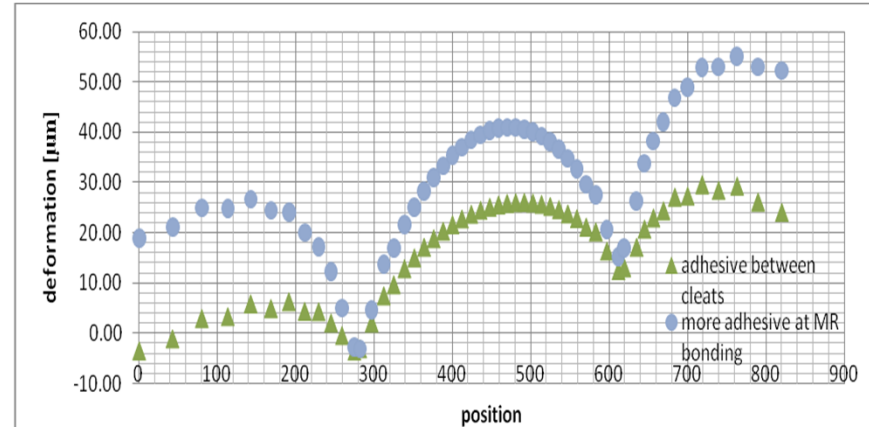
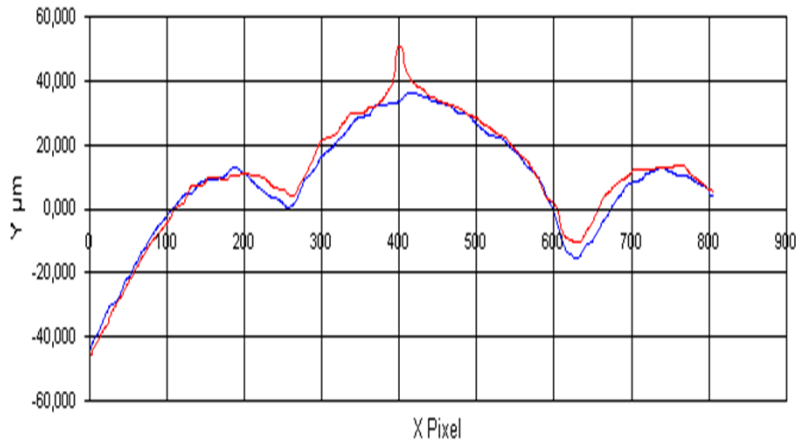
**TED Test**



?



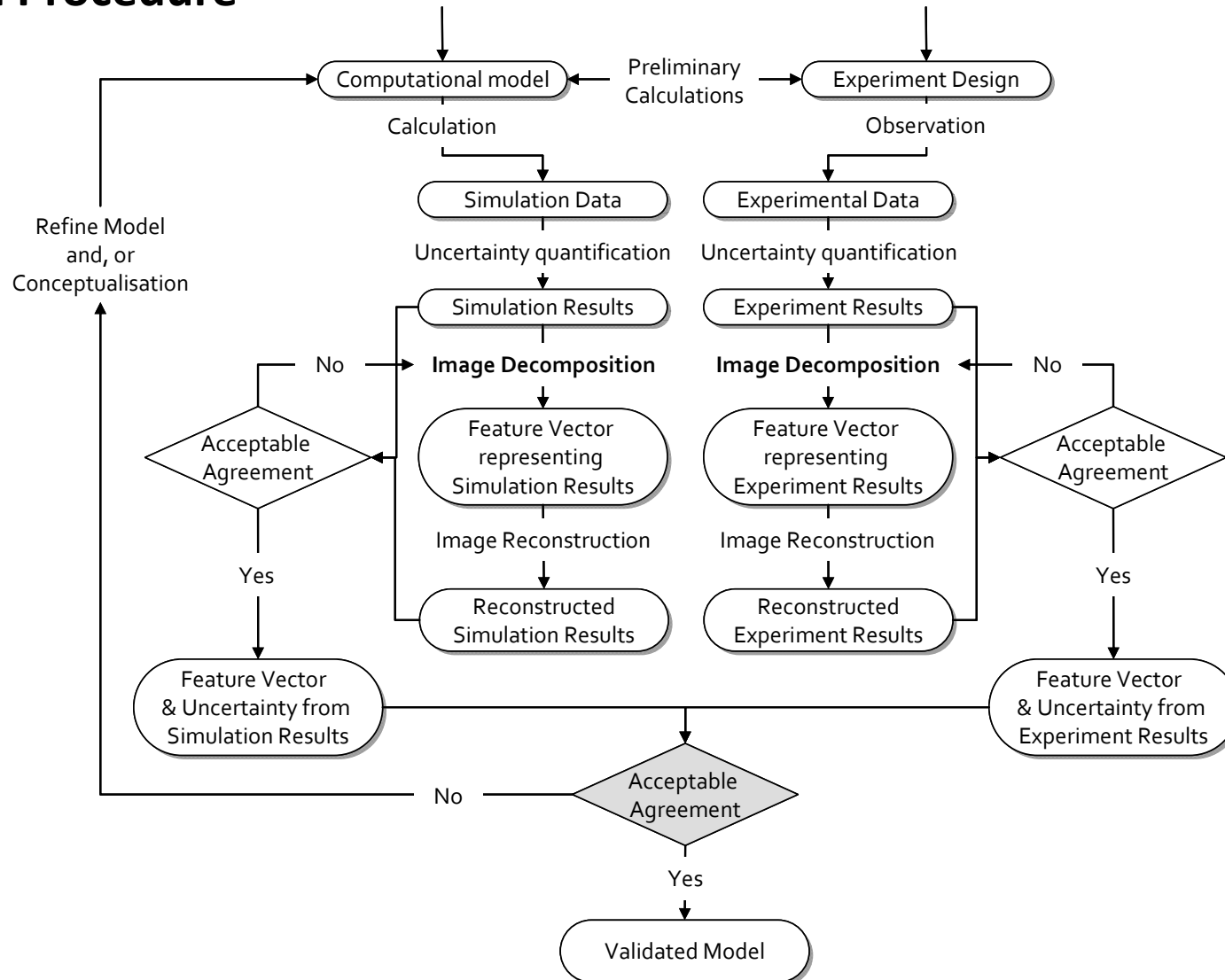
**FEM**



**Qualitative comparison!**

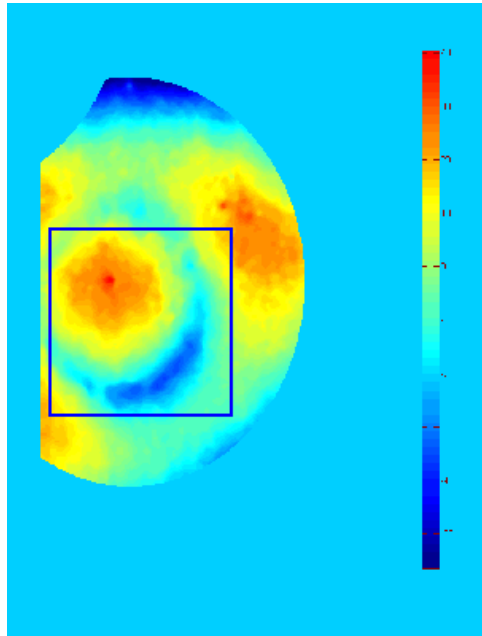


## Validation Procedure

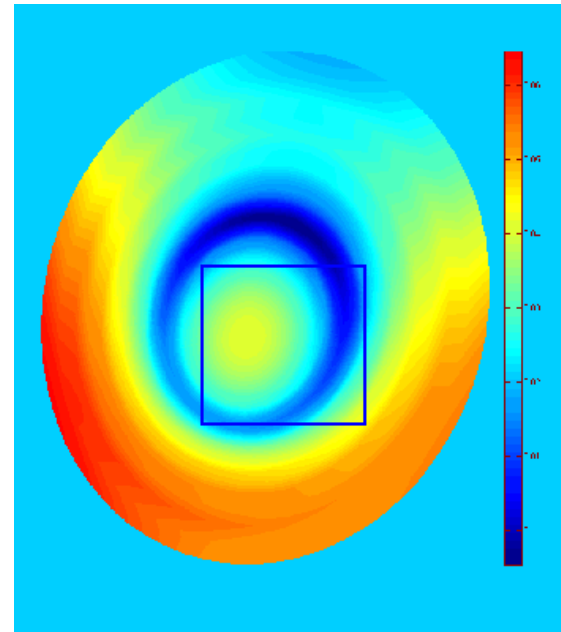


## Validation Procedure (STANT)

**TED Test**

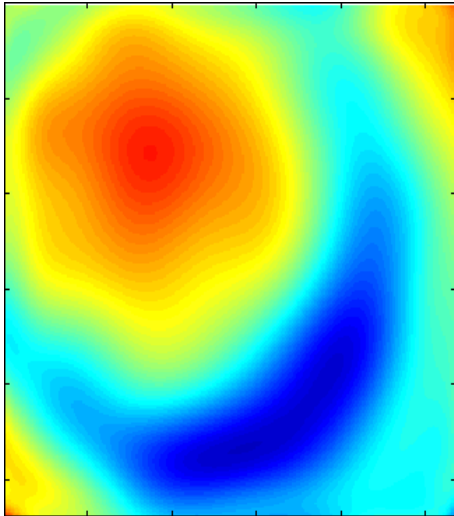


**FEM**

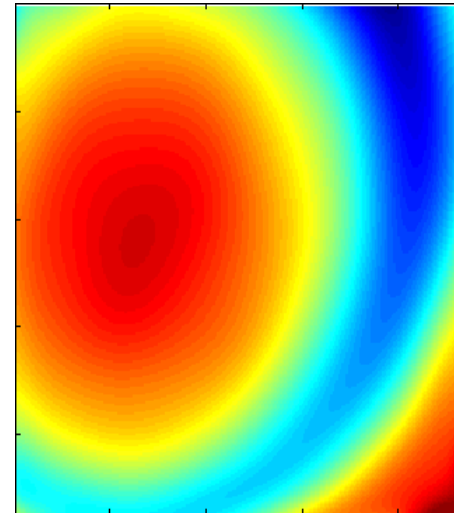


Selection of region of interest to be used for image decomposition

## Validation Procedure (STANT)

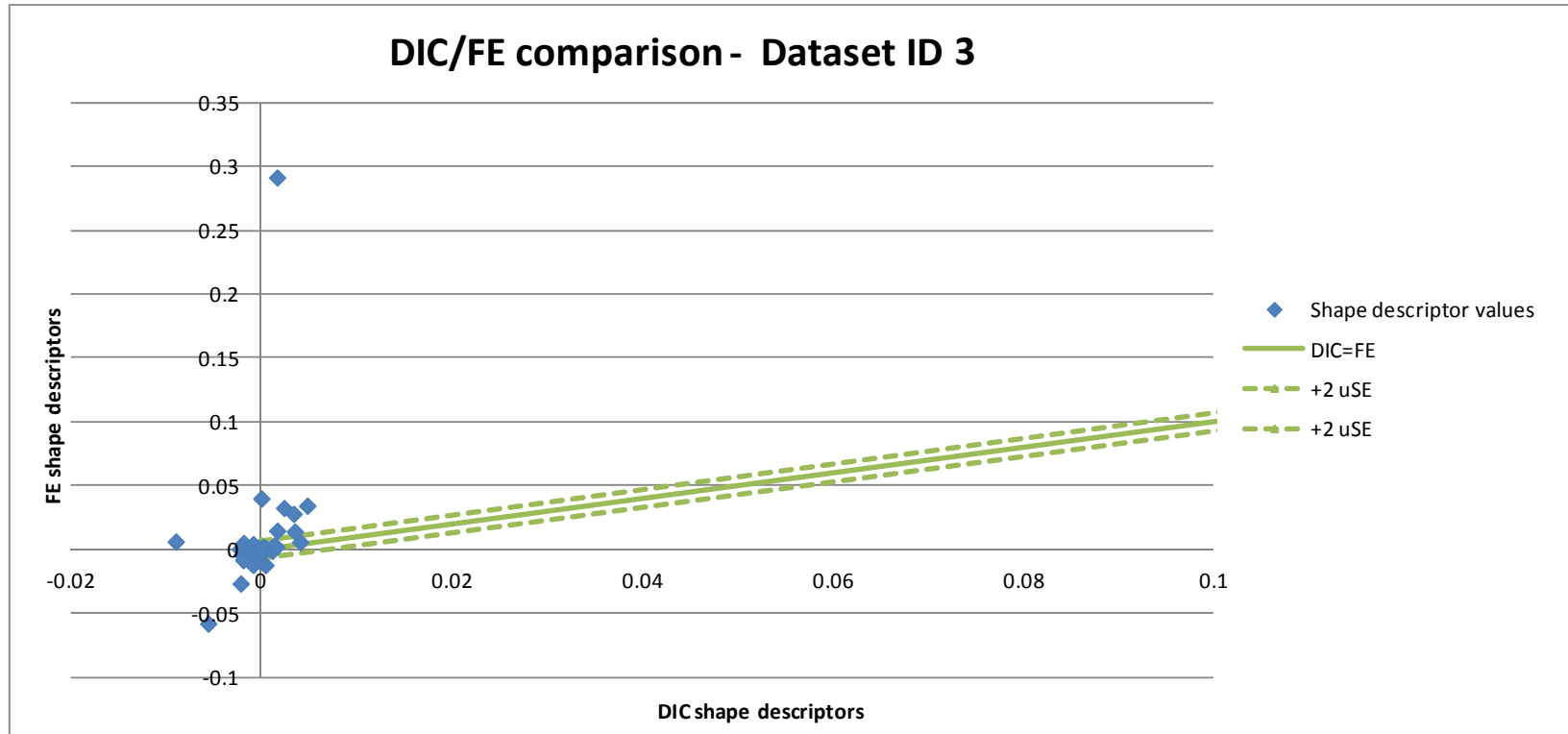


**Decomposition of  
ESPI  
UE=0.002**



**Decomposition of  
FEM  
UM=0.006**

## Validation Procedure (STANT)



**Ucal=0.003**

**UE=0.002**

**USE=0.0036**

$$U(s_E) = \sqrt{U_{cal}^2(\epsilon) + U_E^2}$$



## Part 4: Conclusion and Outlook



## Conclusion

### Model representation

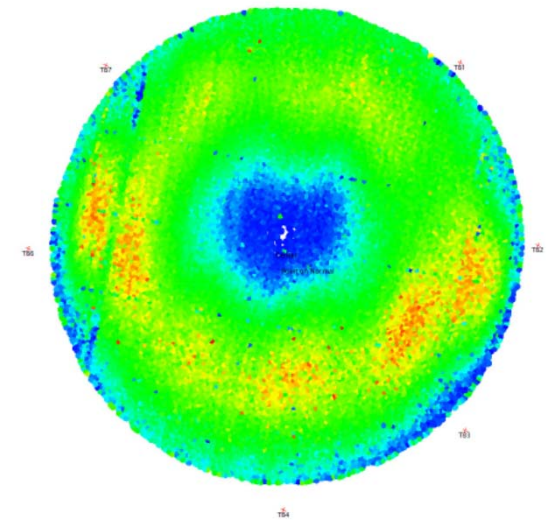
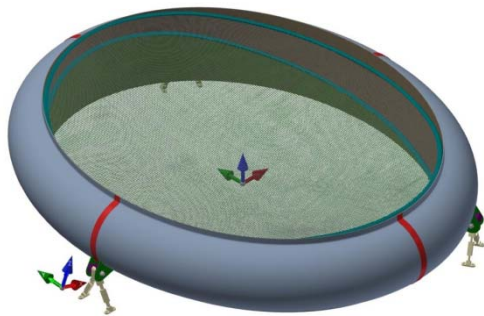
- Reflector deformation is sensitive to bonding layer
- Bonding layer varies across circumference of I/F
- Varying layer is not represented in model
- VANESSA method might help, e.g. in parametric analysis

### Region of interest

- Test was not intended to be used for VANESSA methodology
- No complete test data (some regions missing)
- Orientation and alignment of test data not known
- Test data and FEM data input files don't match (size, maybe orientation, maybe alignment)
- ROI cannot be selected properly, best guess approach

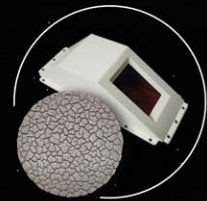
## Outlook

- References on structure would help for ROI selection
- Alignment and orientation should be well known
- Representative/sufficient test data coverage
- Consider for VANESSA methodology during test planning and test activity





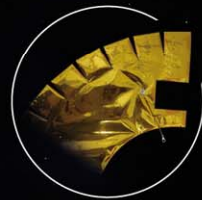
Thank you for your attention.



Launcher and  
Re-entry  
Components



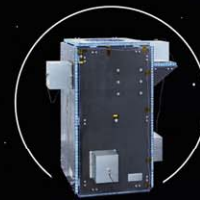
Equipment,  
Instruments



MLI



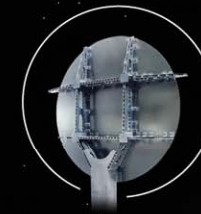
Radiators



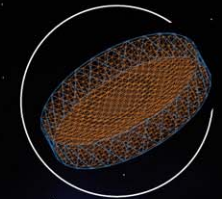
Satellite  
Structures



Antennas



Reflectors



Deployable  
Structures